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HW #2 (Solutions)

(pg 1)

Problem 1

$$(a) \quad \mathcal{P} = \begin{pmatrix} E/c \\ \vec{p} \end{pmatrix}$$

$$(b) \quad \vec{p} = \gamma m \vec{v} \quad ; \quad E = \gamma m c^2 \quad \& \quad \frac{|\vec{p}|}{E} = \frac{v}{c^2} \Rightarrow \boxed{\vec{v} = \frac{\vec{p}}{E}}$$

$$(c) \quad \mathcal{P}_{\pi} = \mathcal{P}_{\mu} + \mathcal{P}_{\nu}$$

$$\Rightarrow \mathcal{P}_{\nu} = \mathcal{P}_{\pi} - \mathcal{P}_{\mu}$$

Now taking a square of both side

$$\mathcal{P}_{\nu}^2 = \mathcal{P}_{\pi}^2 + \mathcal{P}_{\mu}^2 - 2 \mathcal{P}_{\pi} \cdot \mathcal{P}_{\mu}$$

in the CM-frame $\mathcal{P}_{\pi} = \begin{pmatrix} m_{\pi} \\ 0 \end{pmatrix}$. Hence & also

$$\mathcal{P}_{\nu}^2 = 0, \quad \mathcal{P}_{\pi}^2 = m_{\pi}^2 \quad \& \quad \mathcal{P}_{\mu}^2 = m_{\mu}^2$$

Which gives me,

$$0 = m_{\pi}^2 + m_{\mu}^2 - 2 m_{\pi} E_{\mu}$$

$$\Rightarrow E_{\mu} = \frac{m_{\pi}^2 + m_{\mu}^2}{2 m_{\pi}} \quad \dots \dots \dots (1)$$

But $E_{\mu} = m_{\pi} = E_{\mu} + E_{\nu}$ (E conservation)

& $\vec{P}_{\mu} + \vec{P}_{\nu} = 0$ (Momentum conservation)

& $\vec{p}_0 = \hat{z} E_0$

then,

$|\vec{p}_\mu| = |\vec{p}_\nu| \Rightarrow E_0 = m_\pi - E_\mu = \frac{m_\pi^2 - m_\mu^2}{2m_\pi}$

So $v = \frac{|\vec{p}_\mu|}{E_\mu} = \frac{m_\pi^2 - m_\mu^2}{m_\pi^2 + m_\mu^2}$

$\Rightarrow v = \frac{m_\pi^2 - m_\mu^2}{m_\pi^2 + m_\mu^2} c$

(d)

$d_{non-rel} = v \tau$
 $= 0.998 \times 3 \times 10^8 \times 2.2 \times 10^{-6} \text{ m}$
 $= 658.68 \text{ m}$

$v = 0.998c$

$d_{relativistic} = \gamma v \tau$
 $\approx 10.4 \text{ km}$

$\gamma = \left(1 - \frac{v^2}{c^2}\right)^{-1/2}$
 ≈ 15.8

Problem 2 The quickest way to solve this one is

in CM-frame $\mathcal{P}_{\text{tot}} = \mathcal{P}_1 + \mathcal{P}_2 = \begin{pmatrix} E+E \\ (E-E)\hat{z} \end{pmatrix} = \begin{pmatrix} 2E \\ 0 \end{pmatrix}$

in rest-frame,

$$\mathcal{P}'_{\text{tot}} = \begin{pmatrix} m \\ 0 \end{pmatrix} + \begin{pmatrix} E' \\ \vec{p}' \end{pmatrix} = \begin{pmatrix} E'+m \\ \vec{p}' \end{pmatrix}$$

then, invariance of the square of four momentum tells us.

$$\mathcal{P}_{\text{tot}}^2 = \mathcal{P}'_{\text{tot}}{}^2$$

$$\Rightarrow 4E^2 = (E'+m)^2 - |\vec{p}'|^2 \quad \text{--- (1)}$$

Now $|\vec{p}'|^2 = E'^2 - m^2$

Thus (1) \Rightarrow

$$4E^2 = E'^2 + m^2 + 2E'm - E'^2 + m^2$$

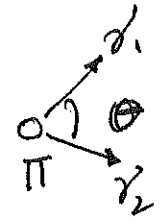
$$\Rightarrow 4E^2 = 2m(E'+m)$$

Now using $T = E - m$ & $T' = E' - m$ we get

$$\boxed{T' = 4T \left(1 + \frac{T}{2m}\right)}$$

for $T \ll m \rightarrow T' \approx 4T$
as expected.

Problem 2.3



$$p_{\pi} = p_1 + p_2$$

$$\Rightarrow p_{\pi}^2 = p_1^2 + p_2^2 + 2p_1 \cdot p_2$$

$$\Rightarrow m_{\pi}^2 = 0 + 0 + 2(E_1 E_2 \cancel{v_1 \cdot v_2} - E_1 E_2 \cos \theta)$$

$$= 2E_1 E_2 (1 - \cos \theta)$$

$$\Rightarrow \boxed{E_1 E_2 = \frac{m_{\pi}^2}{2(1 - \cos \theta)}}$$

Note the above relationship is valid for any reference frame.

the happens when $\theta = \pi$ then.

$$\sqrt{E_1 E_2} = \frac{m_{\pi}}{2}$$

$E_{\gamma} = m_{\pi}/2$ is only ~~valid~~ valid in ~~rest~~ CM frame.

Problem 2.4
observed

$$\frac{\lambda_o}{\lambda_e} = z + 1;$$

emitted.

In this problem the redshift is occurring due to the gravitational effect.

So

$$\frac{\lambda_o}{\lambda_e} = z + 1 = \left(1 - \frac{2GM}{r}\right)^{-1/2}$$

Now on the surface of the Sun the "observed" wavelength is λ_s then.

$$\frac{\lambda_s}{\lambda} = \left(1 - \frac{2GM_\odot}{R_\odot}\right)^{-1/2}$$

Similarly on the Earth's surface

$$\frac{\lambda_E}{\lambda} = \left(1 - \frac{2GM_E}{R_E}\right)^{-1/2}$$

then,

$$\frac{\lambda_s}{\lambda_E} = \frac{\lambda_s/\lambda}{\lambda_E/\lambda} = \frac{\left(1 - \frac{2GM_E}{R_E}\right)^{1/2}}{\left(1 - \frac{2GM_\odot}{R_\odot}\right)^{1/2}}$$

~~20.0.08.12~~



Now remembering to put the c^2 in the correct place

$$\frac{\lambda_E}{\lambda_s} = \left(\frac{1 - \frac{2G_1 M_0}{c^2 R_0}}{1 - \frac{2G_2 M_E}{c^2 R_E}} \right)^{1/2}$$

then

$$\frac{\Delta \lambda}{\lambda_s} = 1 - \frac{\lambda_E}{\lambda_s} \approx 2.12 \times 10^{-6} //$$

Problem 2.5 See ~~book~~ back of the book.

~~2.6~~
Problem 2.6 The radius of the ~~satellite~~ satellite orbit is r then

$$T^2 = \frac{4\pi^2}{G_1 M_E} r^3$$

$$\Rightarrow r = \left(\frac{G_1 M_E}{4\pi^2} \right)^{1/3} T^{2/3} = 2.8 \times 10^7 \text{ m}$$

$$\begin{aligned} T &= 12 \text{ h} = 12 \times 3600 \text{ s} \\ G_1 &= 6.673 \times 10^{-11} \text{ m}^3 \text{ kg}^{-1} \text{ s}^{-2} \\ M_E &= 5.98 \times 10^{24} \text{ kg} \end{aligned}$$

then $v = \frac{2\pi r}{T} \approx 4072.43 \text{ m/s}$

(a) Now clock drifts due to time dilation

$\Delta t' = \delta \Delta t_{GPS}$

$\Rightarrow \Delta t_{GPS} = \Delta t' / \gamma = (1 - \frac{v^2}{c^2})^{1/2} \Delta t' \approx (1 - \frac{1}{2} \frac{v^2}{c^2}) \Delta t'$

then the amount of shift due to SR is

$$-\frac{v^2}{2c^2} \approx -0.92 \times 10^{-10} \text{ s/s}$$

(b) ~~$\Delta t' = \sqrt{1 + \frac{2\Delta\phi}{c^2}} \Delta t_{GPS}$~~

$\Delta t' = \sqrt{1 + \frac{2\Delta\phi}{c^2}} \Delta t_{GPS}$

$\Rightarrow \Delta t_{GPS} \approx (1 - \frac{\Delta\phi}{c^2}) \Delta t'$

Here $\Delta\phi = \phi(R_e) - \phi(r)$ & $\phi(a) = -\frac{GM_e}{a}$

then time shift

$$- \left\{ -\frac{GM_e}{c^2 R_e} + \frac{GM_e}{c^2 r} \right\} = +5.4 \times 10^{-10} \text{ s/s}$$

(c) Net shift is 4.48×10^{-10} s/s = 3.87×10^{-5} s/day

day =

~~24 hrs~~

Then the net error in distance is

$$3.87 \times 10^{-5} \times 3 \times 10^8 \text{ m/day} = 11.6 \times 10^3 \text{ m/day.}$$