PHYS 116C Practice Midterm Solutions

Logan A. Morrison

April 30, 2018

1 Problem One

Consider a object orbiting around the earth at a constant altitude and constant latitude, i.e. orbiting at a constant radius R from the earth's center and a constant angle from the north pole θ_0 . Show that the velocity and acceleration are given by

$$
\frac{d\mathbf{s}}{dt} = R\sin\theta_0 \dot{\phi} \mathbf{e}_{\phi} \tag{1.1}
$$

$$
\frac{d^2\mathbf{s}}{dt^2} = R\sin\theta_0 \left(-\sin\theta_0 \dot{\phi}^2 \mathbf{e}_{\rho} - \cos\theta_0 \dot{\phi}^2 \mathbf{e}_{\theta} + \ddot{\phi} \mathbf{e}_{\phi} \right) \tag{1.2}
$$

Here we are using

$$
x = \rho \sin \theta \cos \phi \tag{1.3}
$$

$$
y = \rho \sin \theta \sin \phi \tag{1.4}
$$

$$
z = \rho \cos \theta \tag{1.5}
$$

1.1 Solution One

First, let's compute $e_{\rho}, e_{\theta}, e_{\phi}$. Given the relationships in [Eq. \(1.3\)](#page-0-0) - [Eq. \(1.5\),](#page-0-1) we find that

$$
dx = (\sin \theta \cos \phi) d\rho + (\rho \cos \theta \cos \phi) d\theta + (-\rho \sin \theta \sin \phi) d\phi \qquad (1.6)
$$

$$
dy = (\sin \theta \sin \phi) d\rho + (\rho \cos \theta \sin \phi) d\theta + (\rho \sin \theta \cos \phi) d\phi \qquad (1.7)
$$

$$
dz = (\cos \theta) d\rho + (-\rho \sin \theta) d\theta + (0) d\phi \qquad (1.8)
$$

Given that

$$
ds = \hat{x}dx + \hat{y}dy + \hat{z}dz = \mathbf{e}_{\rho}d\rho + \rho\mathbf{e}_{\theta}d\theta + \rho\sin\theta\mathbf{e}_{\phi}d\phi
$$
 (1.9)

We can see that

$$
ds = \hat{x} \left(\sin \theta \cos \phi d\rho + \rho \cos \theta \cos \phi d\theta - \rho \sin \theta \sin \phi d\phi \right)
$$
 (1.10)

$$
+ \hat{y} \left(\sin \theta \sin \phi d\rho + \rho \cos \theta \sin \phi d\theta + \rho \sin \theta \cos \phi d\phi \right) \tag{1.11}
$$

$$
+ \hat{z} \left(\cos \theta d\rho + -\rho \sin \theta d\theta \right) \tag{1.12}
$$

Hence, the unit vectors in spherical coordinates are

$$
\mathbf{e}_{\rho} = \hat{x}\sin\theta\cos\phi + \hat{y}\sin\theta\sin\phi + \hat{z}\cos\theta\tag{1.13}
$$

$$
\mathbf{e}_{\theta} = \hat{x}\cos\theta\cos\phi + \hat{y}\cos\theta\sin\phi - \hat{z}\sin\theta \tag{1.14}
$$

$$
\boldsymbol{e}_{\phi} = -\hat{x}\sin\phi + \hat{y}\cos\phi\tag{1.15}
$$

Next, we can compute the velocity by dividing ds by dt :

$$
\frac{d\mathbf{s}}{dt} = \mathbf{e}_{\rho}\dot{\rho} + \rho \mathbf{e}_{\theta}\dot{\theta} + \rho \sin \theta \mathbf{e}_{\phi}\dot{\phi}
$$
 (1.16)

Since $\rho = R = \text{constant}$ and $\theta = \theta_0 = \text{constant}$, we find that

$$
\frac{d\mathbf{s}}{dt} = R\sin\theta_0 \mathbf{e}_{\phi}\dot{\phi}
$$
 (1.17)

Next, to compute the acceleration, we need to take the derivative of ds/dt . In order to do so, we need to compute $d\mathbf{e}_{\phi}/dt$. This is

$$
\frac{d\mathbf{e}_{\phi}}{dt} = \frac{d}{dt} \left(-\hat{x}\sin\phi + \hat{y}\cos\phi \right) = -\left(\hat{x}\cos\phi + \hat{y}\sin\phi \right) \dot{\phi}
$$
(1.18)

Let's try to write this in terms of e_{ρ} and e_{θ} . It isn't to hard to see that

$$
\sin \theta \mathbf{e}_{\rho} + \cos \theta \mathbf{e}_{\theta} = \cos \phi \hat{x} + \sin \phi \hat{y} \tag{1.19}
$$

Thus, we can see that

$$
\frac{d\boldsymbol{e}_{\phi}}{dt} = -\left(\sin\theta\boldsymbol{e}_{\rho} + \cos\theta\boldsymbol{e}_{\theta}\right)\dot{\phi} \tag{1.20}
$$

Now, the acceleration is

$$
\frac{d^2\mathbf{s}}{dt^2} = R\sin\theta_0 \left(\dot{\phi}\frac{d\mathbf{e}_{\phi}}{dt} + \ddot{\phi}\mathbf{e}_{\phi}\right) = R\sin\theta_0 \left(-\dot{\phi}^2\sin\theta_0\mathbf{e}_{\rho} - \dot{\phi}^2\cos\theta_0\mathbf{e}_{\theta} + \ddot{\phi}\mathbf{e}_{\phi}\right) \tag{1.21}
$$

2 Problem Two

A helicopter hovering over a target releases a payload of mass m from rest. Including linear air-drag, the differential equation describing the motions of the payload in the vertical direction is

$$
m\frac{d^2y}{dt^2} = -b\frac{dy}{dt} - mg\tag{2.1}
$$

Solve for the position as a function of time by first solving for the velocity as a function of time. After computing $v(t) = \frac{dy}{dt}$ $\frac{dy}{dt}$, integrate the velocity to compute the position $y(t)$. Using $v(t)$, determine the terminal velocity of the payload, i.e. the limit of the velocity as $t\to\infty$.

2.1 Solution Two

The differential equation for the velocity is given by

$$
\frac{dv}{dt} + \frac{b}{m}v = -g\tag{2.2}
$$

This is solved by use of an integrating factor equal to $I(t) = e^{bt/m}$. Multiplying by $I(t)$ our differential equation can be written as

$$
\frac{d}{dt}\left(e^{bt/m}v(t)\right) = -ge^{bt/m} \tag{2.3}
$$

Integrating both sides, and solving for v we find

$$
v(t) = -\frac{gm}{b} + Ce^{-bt/m}
$$
\n(2.4)

Given that $v(0) = 0$, we find that $C = gm/b$. Hence,

$$
v(t) = \frac{gm}{b} \left(e^{-bt/m} - 1 \right)
$$
 (2.5)

Next, let's integrate this to obtain $y(t)$:

$$
y(t) = \frac{gm}{b} \left(-\frac{m}{b} e^{-bt/m} - t \right) + \mathcal{D}
$$
\n(2.6)

Setting $y(0) = y_0$, we find that

$$
\mathcal{D} = y_0 + \frac{gm^2}{b^2} \tag{2.7}
$$

Therefore,

$$
y(t) = y_0 + \frac{gm^2}{b^2} \left(1 - \frac{b}{m} t - e^{-bt/m} \right)
$$
 (2.8)

Lastly, we can take the limit as $t \to \infty$ of $v(t)$ to find the terminal velocity:

$$
\lim_{t \to \infty} v(t) = -\frac{gm}{b} \tag{2.9}
$$

Notice that this agrees with setting $dv/dt = 0$ in the differential equation for $v(t)$.

3 Problem Three

A very useful identity in quantum field theory and group theory is the Jacobi identity. A manifestation of the Jacobi identity in terms of Levi-Civita symbols is as follows:

$$
\epsilon_{ade}\epsilon_{bcd} + \epsilon_{bde}\epsilon_{cad} + \epsilon_{cde}\epsilon_{abd} = 0 \tag{3.1}
$$

Prove that this identity holds.

3.1 Solution Three

To prove this identity, we use

$$
\epsilon_{iab}\epsilon_{icd} = \delta_{ac}\delta_{bd} - \delta_{ad}\delta_{bc} \tag{3.2}
$$

Before we start using this identity, let's get all repeated indices into the first position by using the anti-symmetry property of the levi-civita symbol (i.e. $\epsilon_{...i...j...} = -\epsilon_{...j...i...}$):

$$
\epsilon_{ade}\epsilon_{bcd} + \epsilon_{bde}\epsilon_{cad} + \epsilon_{cde}\epsilon_{abd} = -\epsilon_{dae}\epsilon_{dbc} - \epsilon_{dbe}\epsilon_{dca} - \epsilon_{dce}\epsilon_{dab} \tag{3.3}
$$

Next, let's use the above identity everywhere

$$
\epsilon_{ade}\epsilon_{bcd} + \epsilon_{bde}\epsilon_{cad} + \epsilon_{cde}\epsilon_{abd} = -(\delta_{ab}\delta_{ec} - \delta_{ac}\delta_{ed}) - (\delta_{dc}\delta_{ea} - \delta_{ba}\delta_{ec}) - (\delta_{ca}\delta_{eb} - \delta_{cb}\delta_{ea}) \quad (3.4)
$$

$$
= -\delta_{ab}\delta_{ec} + \delta_{ac}\delta_{ed} - \delta_{dc}\delta_{ea} + \delta_{ba}\delta_{ec} - \delta_{ca}\delta_{eb} + \delta_{cb}\delta_{ea} \tag{3.5}
$$

$$
=0 \tag{3.6}
$$

4 Problem Four

Prove the following identities:

$$
\nabla(\mathbf{A} \cdot \mathbf{B}) = (\mathbf{A} \cdot \nabla)\mathbf{B} + (\mathbf{B} \cdot \nabla)\mathbf{A} + \mathbf{A} \times (\nabla \times \mathbf{B}) + \mathbf{B} \times (\nabla \times \mathbf{A}) \tag{4.1}
$$

$$
\nabla \times (\mathbf{A} \times \mathbf{B}) = \mathbf{A}(\nabla \cdot \mathbf{B}) - \mathbf{B}(\nabla \cdot \mathbf{A}) + (\mathbf{B} \cdot \nabla)\mathbf{A} - (\mathbf{A} \cdot \nabla)\mathbf{B}
$$
(4.2)

[Hint: Recall that $\epsilon_{iab}\epsilon_{icd} = \delta_{ac}\delta_{bd} - \delta_{ad}\delta_{bc}$]

4.1 Solution Four

Let's work with the first identity first. In tensor notation, this can be written as

$$
\nabla \left(\mathbf{A} \cdot \mathbf{B} \right)_a = B_b \partial_a A_b + A_b \partial_a B_b = \delta_{ac} \delta_{bd} \left(B_b \partial_c A_d + A_b \partial_c B_d \right) \tag{4.3}
$$

Recall that $\epsilon_{iab}\epsilon_{icd} = \delta_{ac}\delta_{bd} - \delta_{ad}\delta_{bc}$. We can use this identity to write

$$
\delta_{ac}\delta_{bd} = \delta_{ad}\delta_{bc} + \epsilon_{iab}\epsilon_{icd} \tag{4.4}
$$

Then, we find that

$$
\nabla (\mathbf{A} \cdot \mathbf{B}) = (\delta_{ad}\delta_{bc} + \epsilon_{iab}\epsilon_{icd}) (B_b \partial_c A_d + A_b \partial_c B_d)
$$
(4.5)

$$
= B_b \partial_b A_a + A_b \partial_b B_a + B_b \partial_c A_d \epsilon_{iab} \epsilon_{icd} + A_b \partial_c B_d \epsilon_{iab} \epsilon_{icd}
$$
\n
$$
(4.6)
$$

$$
= (\boldsymbol{B} \cdot \nabla) \boldsymbol{A} + (\boldsymbol{A} \cdot \nabla) \boldsymbol{B} + B_b (\nabla \times \boldsymbol{A})_i \epsilon_{iab} + A_b (\nabla \times \boldsymbol{B})_i \epsilon_{iab} \tag{4.7}
$$

$$
= (\boldsymbol{B} \cdot \nabla) \boldsymbol{A} + (\boldsymbol{A} \cdot \nabla) \boldsymbol{B} + \boldsymbol{B} \times (\nabla \times \boldsymbol{A}) + \boldsymbol{A} \times (\nabla \times \boldsymbol{B})
$$
(4.8)

Which is what we set out to prove. Next, let's look at the second identity:

$$
\nabla \times (\boldsymbol{A} \times \boldsymbol{B}) = \epsilon_{iab}\partial_a (\boldsymbol{A} \times \boldsymbol{B})_b = \epsilon_{iab}\epsilon_{bcd}\partial_a (\boldsymbol{A}_c \boldsymbol{B}_d) = \epsilon_{iab}\epsilon_{bcd} (\boldsymbol{A}_c \partial_a \boldsymbol{B}_d + \boldsymbol{B}_d \partial_a \boldsymbol{A}_c) \tag{4.9}
$$

Using the usual levi-civita identity, we find

$$
\nabla \times (\mathbf{A} \times \mathbf{B}) = (\delta_{ic}\delta_{ad} - \delta_{id}\delta_{ac}) (A_c\partial_a B_d + B_d\partial_a A_c)
$$
(4.10)

$$
= A (\nabla \cdot \mathbf{B}) + (\mathbf{B} \cdot \nabla) \mathbf{A} - (\mathbf{A} \cdot \nabla) \mathbf{B} - \mathbf{B} (\nabla \cdot \mathbf{A}) \tag{4.11}
$$

which is what we set out to prove.

5 Problem Five

Consider a short lived, but prolific-breeding species inside a box. Suppose the rate of breeding is proportional to the square of the density of beings inside the box and that the species dies off at a rate of γ . The differential equation for the number of beings as a function of time can be written as

$$
\frac{dN}{dt} = AN^2 - \gamma N\tag{5.1}
$$

Find $N(t)$ given $N(0) = 1$. For what ratio of A/γ does the species remain constant, i.e $N(t) = 1?$

5.1 Solution Five

We notice that this differential equation is separable. It separates to

$$
\frac{dN}{N(AN-\gamma)} = dt \tag{5.2}
$$

To integrate the left-hand-side, we use partial fractions:

$$
\frac{1}{N(AN-\gamma)} = \frac{\alpha}{N} + \frac{\beta}{AN-\gamma} = \frac{N(A\alpha+\beta)-\alpha\gamma}{N(AN-\gamma)}
$$
(5.3)

We can see that $\alpha = -1/\gamma$ and hence $\beta = -A\alpha = A/\gamma$. Therefore,

$$
\int \frac{dN}{N(AN-\gamma)} = \frac{1}{\gamma} \int dN \left(-\frac{1}{N} + \frac{A}{AN-\gamma} \right) = \frac{1}{\gamma} \log \left(\frac{AN-\gamma}{N} \right) \tag{5.4}
$$

We thus have

$$
\frac{1}{\gamma} \log \left(\frac{AN - \gamma}{N} \right) = t + C \tag{5.5}
$$

Exponentiating both sides, we find

$$
\frac{AN-\gamma}{N} = Ce^{\gamma t} \tag{5.6}
$$

Solving for N we find

$$
N(t) = \frac{\gamma}{A - Ce^{\gamma t}}\tag{5.7}
$$

Forcing $N(0) = 1$, we find $C = A - \gamma$. Therefore, our solution is

$$
N(t) = \frac{\gamma}{\gamma e^{\gamma t} + A(1 - e^{\gamma t})} = \frac{\gamma}{(\gamma - A)e^{\gamma t} + A}
$$
(5.8)

We can see that for $\gamma = A$ that $N(t) = 1$.

6 Problem Six

Solve the following differential equations

$$
(D2 + 1)(D2 – 1)y = 0
$$
\n(6.1)

$$
(D3 + D2 - 6D)y = 0
$$
\n(6.2)

where $D = d/dx$.

6.1 Solution Six

To solve these problems, we guess a solution of the form

$$
y(x) = Ae^{\omega x} \tag{6.3}
$$

Plugging this into the first equation, we find

$$
Ae^{\omega x}(\omega^2 + 1)(\omega^2 - 1) = 0
$$
\n(6.4)

For $A \neq 0$, we require that $\omega = \pm i$ or $\omega = \pm 1$. Therefore, our solution is

$$
y(x) = c_1 e^{ix} + c_2 e^{-ix} + c_3 e^x + c_4 e^{-x}
$$
\n(6.5)

We can write this in terms of real solutions as

$$
y(x) = c'_1 \sin(x) + c'_2 \cos(x) + c_3 e^x + c_4 e^{-x}
$$
\n(6.6)

To solve the second equation, we again guess $y = Ae^{\omega x}$. Then we find

$$
Ae^{\omega x}\omega(\omega^2 + \omega - 6) = Ae^{\omega x}\omega(\omega + 3)(\omega - 2) = 0
$$
\n(6.7)

Thus, either $\omega = 0, -3, 2$. Therefore, our solution is

$$
y(x) = c_1 + c_2 e^{-3x} + c_3 e^{2x}
$$
\n(6.8)